

Turbulent-laminar patterns in plane Poiseuille flow

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Turbulent-laminar banded patterns in plane Poiseuille flow are studied via direct numerical simulations in a tilted and translating computational domain using a parallel version of the pseudospectral code Channelflow. 3D visualizations via the streamwise vorticity of an instantaneous and a time-averaged pattern are presented, as well as 2D visualizations of the average velocity field and the turbulent kinetic energy. Simulations for $2300 \geq Re_m \geq 700$ show the gradual development from uniform turbulence to a pattern with wavelength 20 half-gaps at $Re_m \approx 1900$, to a pattern with wavelength 40 at $Re_m \approx 1300$ and finally to laminar flow at $Re_m \approx 800$. These transitions are tracked quantitatively via diagnostics using the amplitude and phase of the Fourier transform and its probability distribution. The propagation velocity of the pattern is approximately that of the mean flux and is a decreasing function of Reynolds number. Examination of the time-averaged flow shows that a turbulent band is associated with two counter-rotating cells stacked in the cross-channel direction and that the turbulence is highly concentrated near the walls. Near the wall, the Reynolds stress force accelerates the fluid through a turbulent band while viscosity decelerates it; advection by the laminar profile acts in both directions. In the center, the Reynolds stress force decelerates the fluid through a turbulent band while advection by the laminar profile accelerates it. These characteristics are compared with those of turbulent-laminar banded patterns in plane Couette flow.

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I. INTRODUCTION: PHENOMENON AND METHODS

The transition to turbulence is one of the least understood phenomena in fluid dynamics. Transitional regimes in wall-bounded shear flows display regular patterns of turbulent and laminar bands which are wide and oblique with respect to the streamwise direction. These patterns have been studied in counter-rotating Taylor-Couette flow¹⁻⁷ and in plane Couette flow⁶⁻¹⁴.

Turbulent-laminar banded patterns have also been observed in plane Poiseuille (channel) flow, numerically by Tsukahara et al.¹⁵⁻¹⁹ and experimentally by Hashimoto et al.²⁰. These authors computed turbulence statistics, Reynolds stresses, energy budgets, the skin friction and the near-wall behavior and two-point correlations for these flows. Tsukahara et al.¹⁷ presented detailed visualizations of the mean flow as well as the effect of turbulent bands on heat transport. Hashimoto et al.²⁰ compared the range of Reynolds numbers, wavelengths

and angles of the turbulent bands obtained experimentally with the numerical results reported by Tsukahara et al.¹⁸. Brethouwer et al.¹⁴ simulated a turbulent-laminar pattern in Poiseuille flow as part of a larger study investigating the effects of damping by Coriolis, buoyancy and Lorentz forces on patterns in transitional flows. The goal of the present paper is to extend this work using methods previously employed to study turbulent-laminar banded patterns in plane Couette flow. In particular, we wish to determine if these patterns can be reproduced in the minimal geometry used in simulations of plane Couette flow⁸⁻¹⁰, and to describe their evolution in time, their propagation velocity, and the balance of forces they entail.

Plane Poiseuille flow is generated by an imposed pressure gradient or an imposed flux between two parallel rigid plates. The usual non-dimensionalization and reference frame for plane Poiseuille flow are such that the plates are motionless, and the unit of length is half the distance between them. Several definitions of the Reynolds number are in use, corresponding to different streamwise velocity scales: Re_c uses the velocity at the center of the channel, Re_m uses the mean velocity across the channel, i.e. the flux, and Re_τ uses the wall shear velocity. One of the standard choices, and that made here, is to impose the flux and to scale the velocity by $3u_{\text{mean}}/2$. For stationary walls, this corresponds to a

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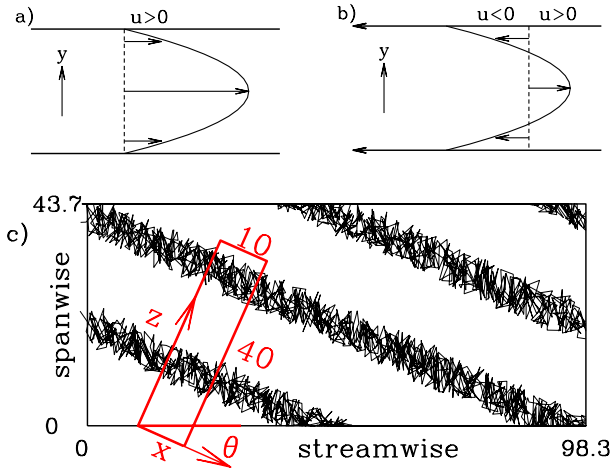


FIG. 1. Plane Poiseuille flow a) with $u_{\text{wall}} = 0$ and $u_{\text{mean}} = 2/3$ and b) with $u_{\text{wall}} = -2/3$ and $u_{\text{mean}} = 0$. c) Tilted reference frame with x aligned with schematically drawn turbulent bands, at an angle of $\theta = 24^\circ$ to the streamwise direction, and z aligned with the pattern wave-vector. The tilted box has dimensions $L_x \times L_z = 10 \times 40$. In order to capture the same pattern, a box aligned with the streamwise and spanwise directions would be required to have dimensions $L_{\text{strm}} \times L_{\text{span}} = 40/\sin(24^\circ) \times 40/\cos(24^\circ) = 98.3 \times 43.7$.

streamwise velocity u satisfying:

$$u_{\text{wall}} = u(y = \pm 1) = 0 \quad (1a)$$

$$u_{\text{mean}} = \frac{1}{2} \int_{-1}^1 dy u(y) = \frac{2}{3} \quad (1b)$$

for which the laminar flow is

$$u_{\text{lam}}(y) = 1 - y^2 \quad (1c)$$

as illustrated in Fig. 1a. In order to follow the patterns more easily, a reference frame is used in which the bounding plates move with velocity $-2/3$. The streamwise mean flow is set to zero for all Re meaning that an appropriate pressure gradient is applied to oppose the moving walls; see Fig. 1b. In this reference frame, the velocity satisfies:

$$u_{\text{wall}} = u(y = \pm 1) = -\frac{2}{3} \quad (2a)$$

$$u_{\text{mean}} = \frac{1}{2} \int_{-1}^1 dy u(y) = 0, \quad (2b)$$

so that the laminar flow becomes

$$u_{\text{lam}}(y) = -\frac{2}{3} + (1 - y^2) \quad (2c)$$

Defining the velocity scale to be

$$\frac{3}{2}(u_{\text{mean}} - u_{\text{wall}}) \quad (3)$$

includes both cases (1a)-(1b) and (2a)-(2b). From (1c) and (2c) it can also be seen that the velocity scale (3) can

be considered to be the difference between the maximum and minimum velocities of the laminar flow corresponding to the imposed flux, i.e. that $Re_m = Re_c$ for the laminar flow. (Many of the references we cite use another factor, one¹⁴ or two¹⁵⁻²¹, in place of the 3/2 in (3); when we cite Reynolds numbers from these references we have multiplied them by the appropriate conversion factor of 3/2 or 3/4.) For the parameter range studied here, the other Reynolds numbers in common use are related to Re_m phenomenologically by $Re_m \sim 1.2Re_c \sim 21Re_\tau$. From now on, unless mentioned otherwise, the Reynolds number Re_m defined from (3) is denoted merely by Re . Our simulations are run by imposing (2a)-(2b); distances are given in units of the half-gap, velocities in units of (3) and times in units of the corresponding advective time.

The reference frame used for this study is not only translated, but also tilted with respect to the usual one, as illustrated in Fig. 1c. As was done for plane Couette flow⁸⁻¹⁰, the horizontal part of the domain is a narrow rectangle whose short direction (here, the x axis, with $L_x = 10$) is parallel to the expected direction of the bands, at an angle of 24° from the streamwise direction. The long direction (here, the z axis, with $L_z = 40$) is parallel to the expected wave-vector of the bands. Thus:

$$\hat{\mathbf{e}}_{\text{strm}} = \cos 24^\circ \hat{\mathbf{e}}_x + \sin 24^\circ \hat{\mathbf{e}}_z, \quad (4a)$$

$$\hat{\mathbf{e}}_{\text{span}} = -\sin 24^\circ \hat{\mathbf{e}}_x + \cos 24^\circ \hat{\mathbf{e}}_z \quad (4b)$$

The reason for choosing 24° in plane Couette flow was that this angle is in the range $[24^\circ, 37^\circ]$ observed experimentally in very large-scale experiments^{6,7} (770 by 340 half-gaps in the streamwise and spanwise directions, respectively), in which the flow was free to choose its own angle; this range of angles is also observed in simulations by Duguet et al.¹² in a domain of similar size. This is also the case for plane Poiseuille flow. Tsukahara et al.¹⁵⁻¹⁷ first produced turbulent-laminar patterns in a domain with dimensions of 51.2 by 22.5 in the streamwise and spanwise directions, leading by construction to $\theta = \tan^{-1}(22.5/51.2) = 23.7^\circ$. The domain used by Brethouwer et al.¹⁴ is very similar (55 by 25) and hence leads to a similar angle of 24.4° . Later simulations¹⁸ in a domain of size 328 by 128 (in which the flow was relatively free to choose its own angle) produced patterns with angles in the range $[20^\circ, 25^\circ]$ while the experiments of Hashimoto et al.²⁰ showed angles in the range $[20^\circ, 30^\circ]$. Our narrow tilted domain enforces an angle of 24° ; only patterns with this angle can be simulated.

The tilted x and z directions are taken to be periodic and y is the usual cross-channel direction, for which boundary conditions (2a) are applied. Periodic boundary conditions on the velocity must be completed by specifying either the constant pressure gradient or the flux (or a combination of the two); we impose zero flux in the spanwise as well as in the streamwise direction. The streamwise, cross-channel and spanwise velocities continue to be denoted by $\mathbf{u} = (u, v, w)$ (even though x, y, z do not correspond to these directions). The domain size is $L_x \times L_y \times L_z = 10 \times 2 \times 40$. The choice of L_z was guided

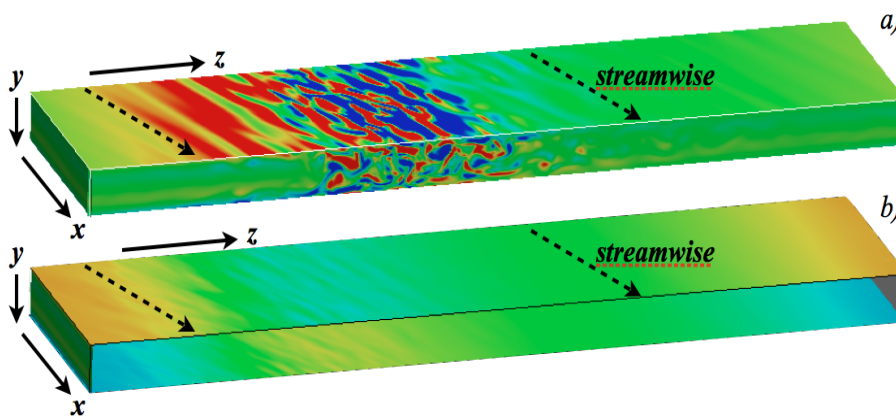


FIG. 2. 3D visualization of the a) instantaneous and b) time-averaged ($\Delta T = 8000$) streamwise vorticity of a turbulent-laminar banded state at $Re = 1100$ over the entire domain of size $L_x \times L_y \times L_z = 10 \times 2 \times 40$. The bands are parallel to the x direction, while the small-scale vortices are aligned in the streamwise direction (dashed arrows) which is oriented at an angle of 24° to the x direction. The maximum and minimum of the instantaneous streamwise vorticity is ± 6.2 while that of the time-averaged vorticity is only ± 0.32 . Features of both are emphasized by choosing the color scale $[-0.5, 0.5]$. The instantaneous vorticity, while strongest on the bounding plates, is present throughout the domain, as shown by the (y, z) plane in front. This plane is omitted from the time-averaged vorticity to reveal contrasting views on the upper and lower plates; the antisymmetry in the vorticity results from the y -reflection symmetry of the average velocity.

by considerations similar to those for the angle, i.e. results from experiments and simulations in plane Couette flow and plane Poiseuille flow. All of the references cited previously^{14–20} reported patterns with wavelengths in the range $[20, 30]$. In our domain, only patterns whose wavelength is a divisor of L_z can be simulated, i.e. 40, 20, 10, etc. The choice $L_x = 10$ is dictated by the requirement that the box be large enough to sustain turbulence, more specifically that the spanwise dimension $L_x \sin \theta$ be wide enough to accommodate a pair of streamwise vortices^{8–10, 22–24}.

Streamwise vortices are indeed a prominent feature of turbulent regions, as shown in the visualization in Fig. 2a of a computed turbulent-laminar pattern. The streamwise vorticity is particularly appropriate for representing turbulence in plane Poiseuille flow since it is zero for laminar flow and, unlike the velocity, is not zero at the plates, near which the turbulence is most intense. The instantaneous vorticity, Fig. 2a, is localized in one region of the domain and is aligned in the streamwise direction. Figure 2b shows the streamwise vorticity averaged over $\Delta T = 8000$ time units. The time-averaged vorticity is much weaker than the instantaneous vorticity, and is antisymmetric under reflection in y , showing that the corresponding velocity is reflection-symmetric in y .

The simulations were performed with a parallelized version of the pseudospectral C++-code Channelflow²⁵. Indeed, Orszag²⁶ first demonstrated the power of his pseudospectral method by calculating the famous critical value $Re_c = 5772$ for loss of stability of plane Poiseuille flow. (It is well known that transition to 3D turbulence takes place at Reynolds numbers well below this 2D instability.) Channelflow’s integration domain is rectangular with periodic boundary conditions in the directions

parallel to the wall and no-slip boundary conditions at the walls. Following Orszag’s original simulations and subsequent work by Kleiser and Schumann²⁷ and Kim et al.²¹, functions are represented by a tensor product of basis functions appropriate for these boundary conditions: Fourier expansions in the wall-parallel directions and Chebyshev polynomials in the wall-normal direction. Functions are also represented on a tensor product of corresponding gridpoints, equally spaced in the wall-parallel directions but on the Chebyshev collocation points in the wall-normal direction. Orszag showed that an expansion in Chebyshev polynomials with Chebyshev collocation points, which are dense near the endpoints, is optimal for resolving fine structures near the walls. In time, a semi-implicit backwards differentiation algorithm²⁸ with a variable time-step is employed. The linear (diffusive) part of the Navier-Stokes equations is carried out implicitly in spectral space, since the Fourier modes are not coupled by the Laplacian. The nonlinear (advective) part is carried out explicitly in the grid space, where multiplications are local.

This version of Channelflow is capable of working in parallel on compute clusters by distributing the flow fields over large numbers of processors, each of which time-steps its share of data. The spectral coefficients of the velocity field along x and z are distributed in such a way that one processor has access to all Chebyshev coefficients of a small patch of (x, z) Fourier modes. Since the linear diffusive step does not couple the different Fourier modes, the implicit part of the equations can be solved directly. For the explicit calculation of the nonlinear advective term, however, a global transpose operation of the distributed data is necessary, which is implemented with routines from the FFTW package²⁹. First, the Cheby-

shev transforms, which do not couple the different Fourier modes, are computed locally. Subsequently, a transpose is performed in the x, y direction and 2D Fourier transforms of $x - z$ planes are computed in parallel for the different values of y . After the explicit calculation of the nonlinear terms, the transformations are performed in reversed order to obtain the spectral representation, which can be added to the linear part.

We use $N_x \times (N_y + 1) \times N_z = 128 \times 65 \times 512 = 4.2 \times 10^6$ points or modes to represent the domain of size $L_x \times L_y \times L_z = 10 \times 2 \times 40$, with a spacing of $\Delta x = \Delta z = 0.08$ and Δy ranging from $\Delta y_{\text{wall}} = 1 - \cos(\pi/64) = 0.001$ to $\Delta y_{\text{mid}} = \cos(31\pi/64) = 0.05$. For the highest Reynolds number we simulate, $Re = 2300$, the ratio between the viscous wall unit and the half-gap is 0.009, so $\Delta x^+ = \Delta z^+ = 0.08/0.009 = 9$, $\Delta y_{\text{wall}}^+ = 0.001/0.009 = 0.11$ and $\Delta y_{\text{mid}}^+ = 0.05/0.009 = 5.6$. For plane Poiseuille flow, it is crucial to have sufficient resolution in the cross-channel direction near the walls, where the turbulence is concentrated. The resolution used here is similar to that used in the simulations by Kim et al.²¹ ($Re = 4200$; $Re_\tau = 180$) and Jiménez & Moin²²; although these authors used $N_y = 128$, they studied Reynolds that were about twice the highest Reynolds number investigated here and stated that using $N_y = 64$ produced similar results²¹. In terms of wall units, our resolution is the same or finer than that used in these studies. Our resolution is also close to that used by Tsukahara et al.^{16,17} for $Re = 1732$ ($Re_\tau = 80$) and $Re = 1327$ ($Re_\tau = 64$), and higher than that of Brethouwer et al.¹⁴, who used $\Delta x^+ = 15$, $\Delta z^+ = 6.7$ and $N_y = 32$ for a case with $Re = 700$ ($Re_\tau = 69$). The timestep varied from $\Delta t = 0.03$ for $Re = 900$ to $\Delta t = 0.015$ for $Re = 2300$. Simulations run on the IBM x3750 of the IDRIS super-computer center using 32 processes typically took about 16 wall clock hours to simulate 10 000 advective time units.

II. REYNOLDS-NUMBER SCAN

Figure 3 shows spatio-temporal diagrams of the spanwise velocity. For $Re \leq 2000$, each simulation is a continuation of the corresponding part of a long simulation in which the Reynolds number is decreased in discrete decrements of 100, which will be presented in Fig. 5 and which itself is initialized with random noise. The simulations with $Re > 2000$ were all initialized with the final state of the $Re = 2000$ run.

The timeseries show $w(z_j, t)$ at $x = 0$, $y = 0.8$ (near the upper plate), for 32 values z_j separated by intervals of $\Delta z = L_z/32$ for Reynolds numbers varying from 2300 down to 800. Dark patches indicate rapid large-amplitude oscillations in the spanwise velocity, i.e. turbulent regions. The surrounding lighter patches are composed of straight lines, indicating locations at which the spanwise velocity remains constant or nearly so, i.e. quasi-laminar regions.

For the highest Reynolds number, $Re = 2300$, the entire interval $0 \leq z \leq L_z$ is dark, indicating turbulence which is statistically uniform over the domain. As Re is lowered to 2100, quiescent patches appear which move towards the left (opposite to $(u_{\text{mean}} - u_{\text{wall}})$). Timeseries for $1900 \geq Re \geq 1400$ show two clearly delineated turbulent bands with a fairly well-defined wavelength and velocity. From $Re = 1300$ to $Re = 1200$, there is a transition from two to one turbulent band. New turbulent patches repeatedly branch off from existing ones; these are more persistent and long-lasting for $Re = 1300$ than for 1200. The velocity of the pattern decreases. For $Re = 1100$, the pattern comprises a single band and is almost stationary; the band begins to disappear at $T = 8700$, becoming completely laminar by $T = 9000$. For $Re = 1000$ and 900, a single right-going band is present, which, for $Re = 900$ disappears at $T = 9700$. For $Re = 800$, the band disappears earlier, at around $T = 2200$.

For plane Couette flow, a qualitative distinction can be made between patterns at higher Re , in which the turbulent and quasi-laminar regions each occupy approximately half of the domain, and patterns at lower Re , in which the turbulent band occupies a smaller fraction of the domain. The averaged high- Re patterns were shown⁹ to have a trigonometric dependence on z . In contrast, the bands at lower Re were shown to be isolated states, in that they retain their size when placed in a wider domain⁸ and are surrounded by truly laminar regions. A comparison of the states at $Re \geq 1400$ with those at $Re < 1300$ shows that this distinction seems also to apply to plane Poiseuille flow. For $Re \geq 1400$, the turbulent bands occupy about half the width of the domain, with the other half consisting of slightly chaotic flow, as shown by the small-scale oscillations in $w(z_j, t)$. For $Re < 1300$, the single turbulent band occupies much less than the width of the domain and the flow reverts to laminar quite close to the boundaries of the band, as shown by the straight lines $w(z_j, t)$ for z_j within a few multiples of $L_z/32$.

The branching events at $Re = 1300$ and $Re = 1200$ indicate bistability between a pattern with wavelength 40 and wavelength 20; a domain with a larger or different L_z would almost surely display bistability at a different value of Re . There is also clearly an important random component in the fact that relaminarisation occurs at $Re = 1100$ but not at $Re = 1000$. Simulations with different initial conditions would almost surely lead to relaminarisation at different times; the properties of these events must be studied statistically, as has been done for pipe flow^{30,31} and for Couette flow³².

Figure 4 shows the propagation velocity as a function of Re . Propagation in the z direction is to be expected, since there is a substantial overlap between the z direction and the streamwise direction. The fact that the speeds are so small demonstrates that the turbulent bands move essentially at the speed of the mean flow, as was noted by Tsukahara et al.¹⁷ and by Hashimoto et al.²⁰. As was seen in Fig. 3, $Re \approx 1100$ separates prop-

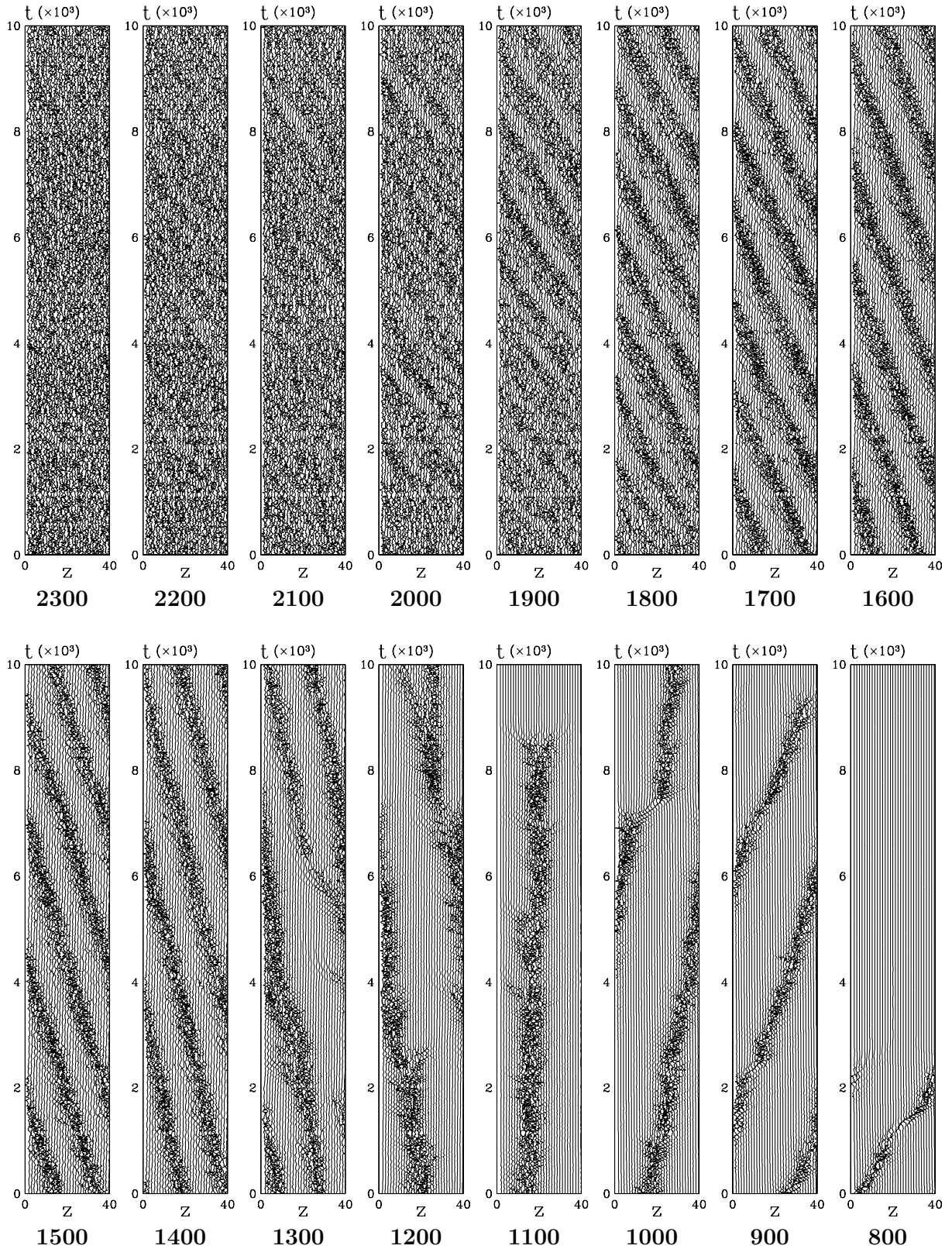


FIG. 3. Spatio-temporal plots from simulations in domain of size $L_x \times L_y \times L_z = 10 \times 2 \times 40$ in which the z direction is oriented along the pattern wave-vector. Spanwise velocity timeseries $w(z_j, t)$ are shown for points along the line $x = 0, y = 0.8$ at 32 equally spaced values $z_j = jL_z/32$ for $2300 \geq Re \geq 800$. Uniform turbulence can be seen for $Re = 2300$, traces of laminar patches for $2200 \geq Re \geq 2000$, increasingly well-defined left-going bands of wavelength 20 for $1900 \geq Re \geq 1400$, a change in wavelength and direction for $Re = 1300, 1200$, and an increasingly weak and fragile right-going pattern of wavelength 40 for $1100 \geq Re \geq 800$.

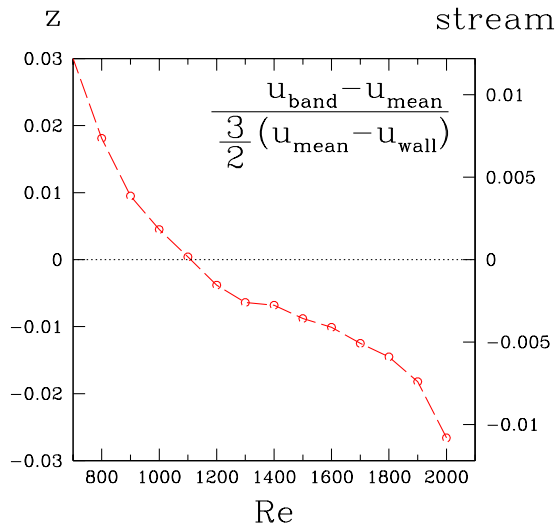


FIG. 4. Propagation speed of turbulent bands with respect to the mean flow. The scale on the left (right) indicates the velocity in the z (streamwise) direction; the two scales are related by the trigonometric factor $\sin 24^\circ$.

agation to the left (slower than the mean flow) and to the right (faster than the mean flow). It is because the pattern at $Re = 1100$ is approximately stationary in the frame of the mean flow that this Reynolds number was chosen to display the time-averaged flow shown in Fig. 2. Although the number of bands is quantized, their velocity is not; moreover the velocity varies smoothly through the change in the number of bands at $Re = 1100$. Therefore it seems likely that the velocity presented in Fig. 4 is independent of the domain. Propagation velocities are in general quite sensitive to resolution; previous simulations with less resolution in y showed the propagation velocity to change sign at $Re = 1400$ instead of $Re = 1100$. We have verified that the velocity does not change appreciably with higher resolution.

An overall view of the evolution of the pattern can be seen in the Reynolds-number scan of Fig. 5. This figure describes a simulation initialized at $Re = 2000$ with a random initial condition and in which the Reynolds number is lowered in discrete steps of 100, remaining at each Re for a time of length $\Delta T = 2000$. (The simulations of length $\Delta T = 10000$ in Fig. 3 are continuations for another $\Delta T = 8000$ of each of the sections of Figure 5). Figure 5a shows streamwise velocity profiles $u(y)$ at intervals of $\Delta T = 2000$ at a fixed value of x and z ; these are flat or parabolic, depending on whether the corresponding flow or region is turbulent or laminar. Figure 5b, like Fig. 3, shows timeseries of the spanwise velocity $w(z_j, t)$ at 32 equally spaced points in z .

The evolution from the random initial condition at $Re = 2000$ leads rapidly to turbulence which is uniform (without bands). As Re is lowered past $Re \approx 1800$, two quiescent patches appear (though the long time series of Fig. 3 show that a muted version of this pattern already appears for higher Re , given sufficient time). Sev-

eral transitions are clearly visible: from two turbulent bands to one at $Re = 1200$, from leftwards to rightwards motion at $Re = 1100$, and from one turbulent band to none at $Re = 700$. “Although we have not checked systematically for hysteresis, a simulation initialized with the single-band $Re = 1100$ solution evolved quickly to a left-moving pattern with two bands when the Reynolds number was increased to 1400.”

Figures 5c,d show that these tendencies can be measured quantitatively via the modulus $|\hat{w}_m(t)|$ and the phase $\hat{z}_m(t)$ of the discrete Fourier transform in z :

$$w(z_j, t) = \sum_m \hat{w}_m(t) e^{imz_j \frac{2\pi}{L_z}} = \sum_m |\hat{w}_m(t)| e^{im(z - \hat{z}_m(t)) \frac{2\pi}{L_z}} \quad (5)$$

averaged over appropriate time intervals:

$$\langle |\hat{w}_m(t)| \rangle_{1000} \equiv \frac{1}{1000} \int_{t'=0}^{1000} dt' |\hat{w}_m(t+t')| \quad (6a)$$

$$\langle \hat{z}_m(t) \rangle_{100} \equiv \frac{1}{100} \int_{t'=0}^{100} dt' \hat{z}_m(t+t') \quad (6b)$$

In Fig. 5c, $\langle |\hat{w}_2(t)| \rangle_{1000}$ rises from a low value when Re is decreased below 1900, and is then overtaken by $\langle |\hat{w}_1(t)| \rangle_{1000}$ at $Re = 1200$, when one of the turbulent bands disappears. Although the pattern for $1200 \geq Re \geq 800$ has wavelength $\lambda = 40$, it also contains higher harmonics and so w_2 remains non-negligible.

Figure 5d shows the averaged phases $\langle \hat{z}_m(t) \rangle_{100}$. When the modulus is small, the phase loses significance, so phases are shown only when the modulus exceeds a heuristically determined threshold, 0.24 for $m = 1$ and 0.4 for $m = 2$. The phases $\langle \hat{z}_2(t) \rangle_{100}$ and $\langle \hat{z}_2(t) \rangle_{100+L_z/2}$ track the centers of the two turbulent bands seen in Fig. 5b for $1800 \geq Re > 1200$, while $\langle \hat{z}_1(t) \rangle_{100}$ tracks the center of the single turbulent band for $1200 \geq Re \geq 800$.

The instantaneous values of $a(t) \equiv |\hat{w}_m(t)|$ collected for each Reynolds number during the time that the flow is partly or entirely turbulent (see Fig. 3) can also be used to construct probability distribution functions. Because $|\hat{w}_m(t)|$ is a modulus, the range $a_{j-1} < |\hat{w}_m(t)| \leq a_j$ corresponds to an annulus in the two-dimensional Cartesian space of $(\hat{w}_m^r, \hat{w}_m^i)$ of area

$$\pi(a_j^2 - a_{j-1}^2) = 2\pi \frac{(a_j + a_{j-1})}{2} (a_j - a_{j-1}) \quad (7)$$

The bin boundaries a_j can be chosen to correspond to annuli of equal size by taking

$$a_j \equiv \sqrt{\frac{j}{N_{\text{bin}}}} \max_t |\hat{w}_m(t)| \quad (8a)$$

$$p_j \propto |\{t : a_{j-1} < |\hat{w}_m(t)| \leq a_j\}| \quad (8b)$$

where $|\{\}|$ denotes the number of elements of a set. Another possibility is to choose bin boundaries a_j which are equally spaced and to correct for the difference in annular areas (7) by dividing p_j by $(a_j + a_{j-1})/2$. A third possibility is to choose bin boundaries such that each bin contains the same number of values, and to divide p_j by $(a_j - a_{j-1})(a_j + a_{j-1})/2$. All three procedures lead to similar probability distribution functions.

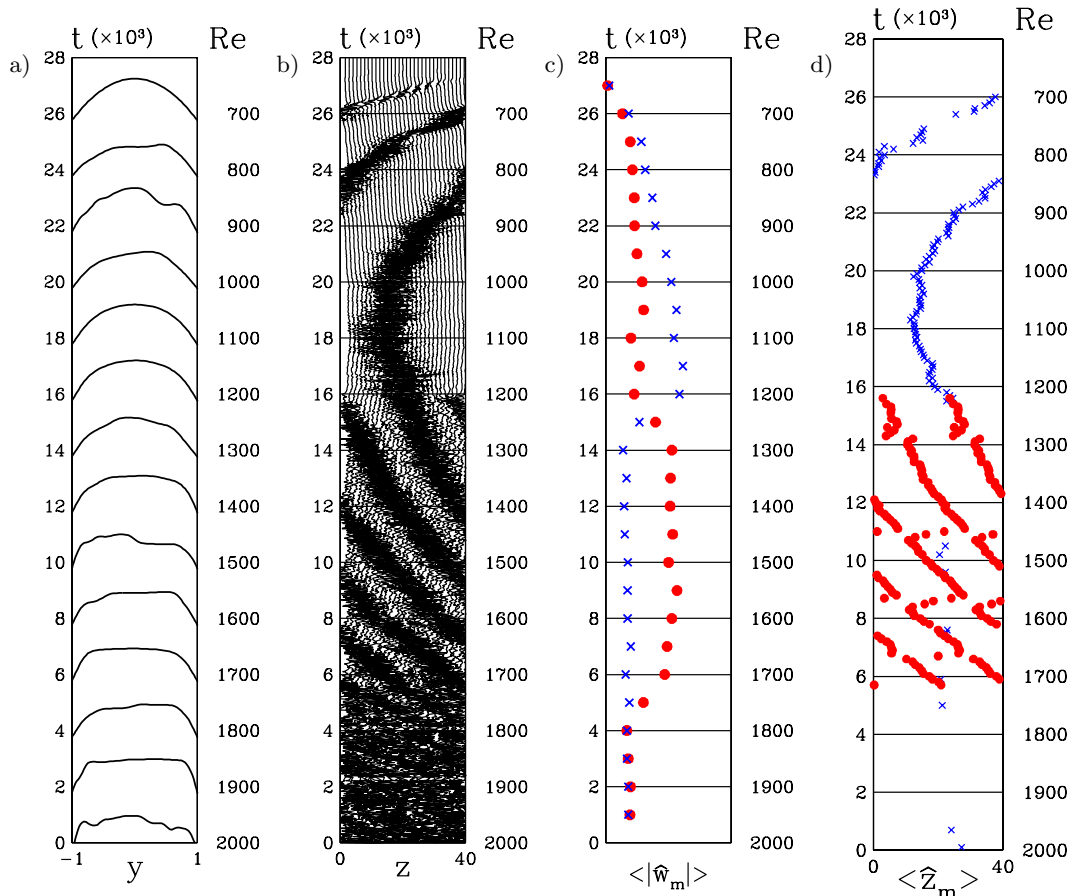


FIG. 5. Simulations in domain of size $L_x \times L_y \times L_z = 10 \times 2 \times 40$. The Reynolds number is decreased from 2000 to 700 in discrete steps at time intervals of $\Delta T = 2000$, as indicated along the axes on the right.

- a) Instantaneous representative streamwise velocity profiles $u(y)$ along the line $x = z = 0$ at intervals of $\Delta T = 2000$.
b) Spanwise velocity timeseries $w(z_j, t)$ along the line $x = 0, y = 0.8$ at 32 equally spaced values $z_j = jL_z/32$.
c) Temporal average $\langle |\hat{w}_m(t)| \rangle_{1000}$ (arbitrary units) of the modulus of the z -Fourier transform \hat{w}_m of the spanwise velocity.
d) Temporal average $\langle \hat{z}_m(t) \rangle_{100}$ of the phase of \hat{w}_m at times for which $\langle |\hat{w}_m(t)| \rangle_{100}$ is sufficiently large.
For c),d), the red disks indicate $m = 2$ ($\lambda = 20$), while the blue crosses indicate $m = 1$ ($\lambda = 40$).

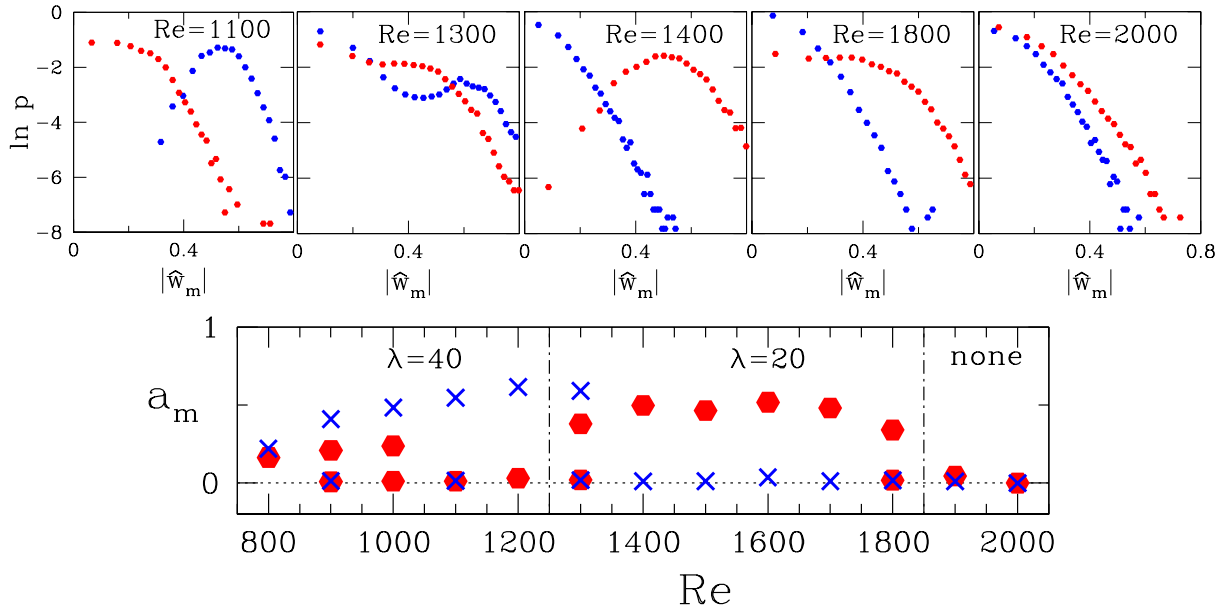


FIG. 6. Above: probability distributions of the moduli of the Fourier components $|\hat{w}_m(t)|$ for $m = 1$ and $m = 2$ for representative values of the Reynolds number. Below: maxima of the PDFs as a function of Reynolds number showing changes in regime at $Re \approx 1900$ and $Re \approx 1300$. The red disks indicate $m = 2$ ($\lambda = 20$), while the blue crosses indicate $m = 1$ ($\lambda = 40$). The maximum of both PDFs is located at zero for $Re \geq 1900$. For $1900 > Re \geq 1300$, the $m = 2$ PDF has a maximum away from zero, indicating a pattern of this wavenumber. For $1300 \geq Re \geq 800$, the $m = 1$ PDF has a maximum away from zero, indicating a pattern of this wavenumber.

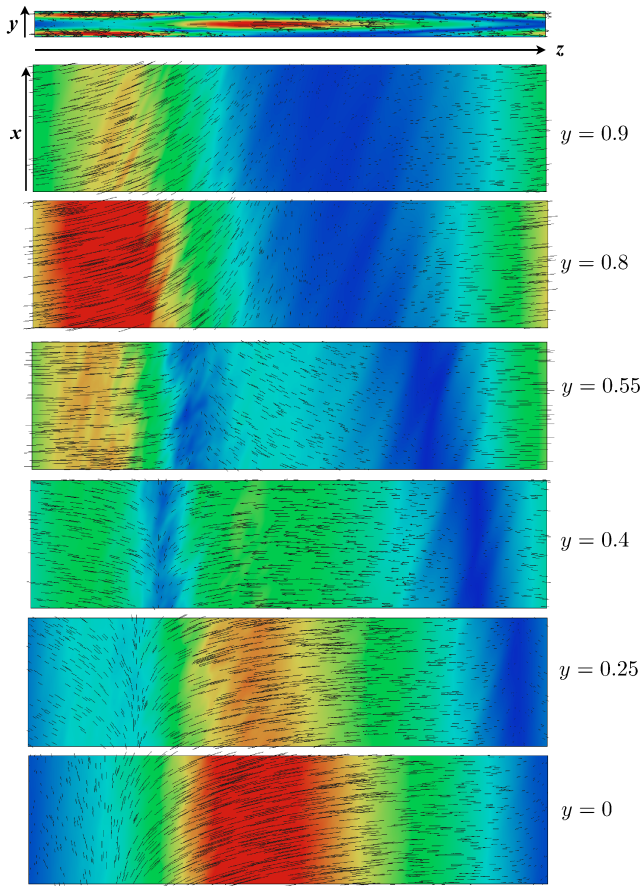


FIG. 7. Time-averaged deviation from laminar flow at $Re = 1100$ on a typical (z, y) plane (top) and on (x, z) planes at various values of y . The (z, y) plot has been stretched by a factor of 3 in the y direction for visibility. Flow for $y < 0$ resembles that for $y > 0$. Arrows indicate the direction of the velocities while colors indicate its magnitude. Scale $[0, 0.2]$.

Figure 6 displays probability distribution functions for $m = 1$ and $m = 2$ for representative Reynolds number values. In the absence of a pattern, in particular for uniform turbulence, the maximum (most probable value) for $|\hat{w}_m(t)|$ is zero, while for patterned flows, the maximum is non-zero. The PDFs yield thresholds:

- $Re \approx 1900$, separating uniform turbulence and a pattern with $m = 2$, i.e. $\lambda = 20$
- $Re \approx 1300$, separating patterns with $m = 2$ and $m = 1$, i.e. $\lambda = 20$ and $\lambda = 40$
- $Re \approx 800$, separating a pattern with $m = 1$ from laminar Poiseuille flow.

III. MEAN FLOW AND FORCE BALANCE

Figure 7 presents views on various planes of the deviation of the time-averaged flow from the laminar velocity. The Reynolds number is 1100, as in Fig. 2. The flow varies a great deal with y and z , but depends little on x , as predicted for the tilted domain. In order to gain more insight into this flow, we therefore form a 2D field by averaging over L_x as well as $\Delta T = 8000$:

$$\langle \mathbf{u} \rangle(y, z) \equiv \int_{t=0}^{8000} dt \int_{x=0}^{L_x} \mathbf{u}(x, y, z, t) \quad (9)$$

where $\langle \rangle$ has been redefined from (6). Fig. 8 presents various aspects of $\langle \mathbf{u} \rangle$. This figure agrees extremely well with Fig. 3 of Tsukahara et al.¹⁷, which shows similar quantities for a patterned flow at $Re = 1327$ averaged over time and L_x . Tsukahara et al.¹⁷ do not show the spanwise velocity, and include a number of other quantities such as the shear stress and Reynolds shear stress not shown here. The field is reflection-symmetric in y . (Reflection in y changes the sign of the cross-channel velocity, as it does for the streamwise vorticity shown in Fig. 2b.)

Figure 8a shows the streamwise velocity. Unlike the other parts of Fig. 8, this subfigure includes the laminar flow. The waviness corresponds to the alternation of parabolic and plug profiles which occur in laminar and turbulent regions, respectively. The cross-channel velocity (Fig. 8b) still shows small-scale features despite the averaging over L_x and $\Delta T = 8000$. The spanwise velocity (Fig. 8c) shows distinctive chevron features. Figure 8d depicts the streamfunction associated with the deviation of the mean velocity from the laminar flow in the (y, z) plane. Two wide counter-rotating cells are stacked in the gap. This is also the form of the mean flow observed experimentally in the presence of a turbulent spot by Lemoult et al.³³. The direction of rotation of these cells is such as to slow the streamwise flow in the middle of the channel and accelerate it near the walls, in effect transforming a parabolic profile to the slug profile.

Figure 8e presents the turbulent kinetic energy defined by

$$E_{\text{turb}} \equiv \frac{1}{2} \langle \tilde{\mathbf{u}} \cdot \tilde{\mathbf{u}} \rangle, \quad \tilde{\mathbf{u}} \equiv \mathbf{u} - \langle \mathbf{u} \rangle \quad (10)$$

which is concentrated very near the boundaries, where the shear of the laminar profile is greatest. The counter-rotating cells (d) are centered at the same value of z as the turbulent kinetic energy (e), but the maximum deviation in the streamwise velocity (a) is located to the right of this location. A shift between these quantities was previously noted by Tsukahara et al.¹⁷ as well as in the case of plane Couette flow^{6,7,9,10}.

We can compare the appearance of these mean quantities with the analogous ones in Fig. 5 of Barkley and Tuckerman⁹ for plane Couette flow. For plane Couette flow, the turbulent kinetic energy occupies most of the

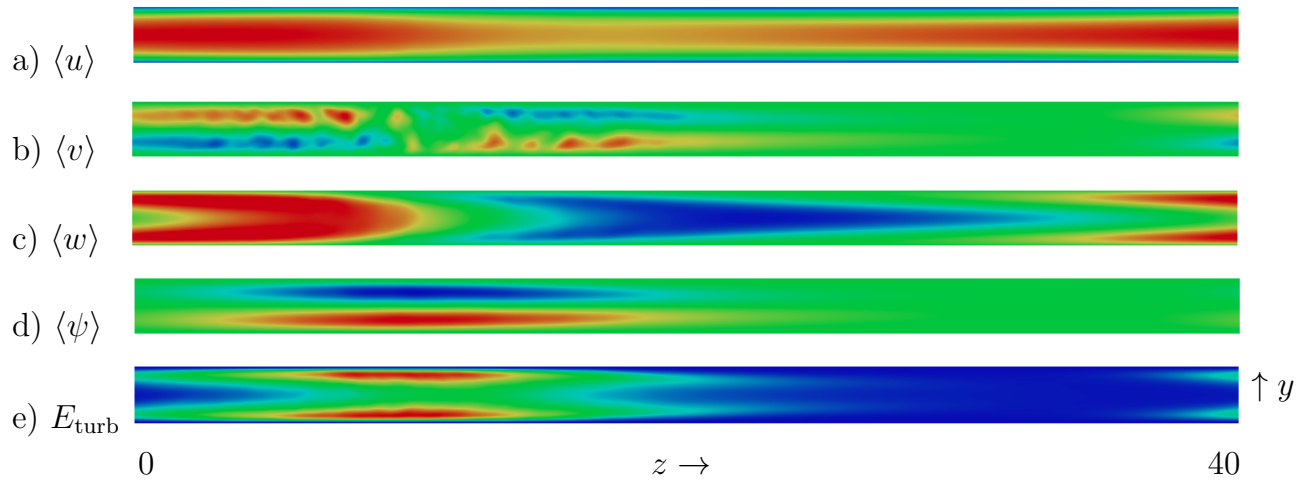


FIG. 8. Mean flow in the (y, z) plane from simulation at $Re = 1100$ averaged over L_x and over time of $\Delta T = 8000$.

- a) Streamwise velocity $\langle u \rangle$ including laminar profile: the undulations correspond to profiles which are slug-like (turbulent regions) or parabolic (laminar regions) in Fig. 5a. Scale $[-0.67, 0.33]$.
 b) Cross-channel velocity $\langle v \rangle$: small-scale structures are still visible despite averaging. Scale $[-0.003, 0.003]$.
 c) Spanwise velocity $\langle w \rangle$ with characteristic chevrons. Scale $[-0.05, 0.05]$.
 d) Streamfunction $\langle \psi \rangle$ shows two superposed layers of cellular flow in the (y, z) plane. The laminar velocity has been subtracted.
 e) Turbulent kinetic energy E_{turb} : the red regions show a strong concentration near the bounding plates. Scale $[0, 0.012]$.

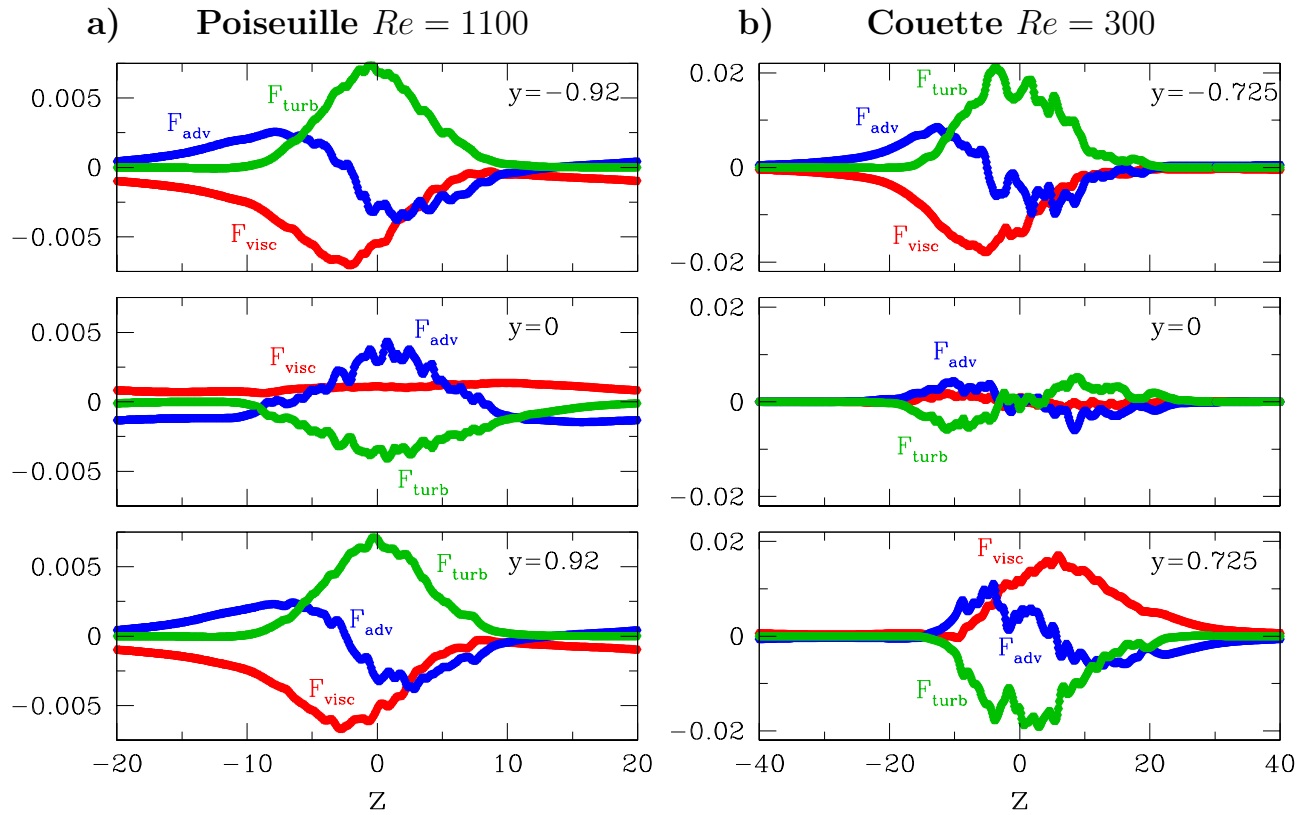


FIG. 9. Streamwise Reynolds stress (green, (12a)), advective (blue, (12b)) and viscous (red, (12c)) forces as a function of z at three y locations for the mean flow associated with turbulent-laminar (a) Poiseuille flow at $Re = 1100$ and (b) Couette flow at $Re = 300$. Near the lower wall ($y < 0$), for both flows, the Reynolds stress force accelerates the fluid through the turbulent band while the viscous force decelerates it; advection by the basic laminar flow changes sign in the middle of the band. In the center ($y = 0$), for Poiseuille flow, advection accelerates the fluid while the Reynolds stress force decelerates it; for Couette flow, this situation is reversed over half the turbulent region. Near the upper wall ($y > 0$), the force balance for Poiseuille flow is identical to that near the lower wall; for Couette flow the balance near the upper and lower walls are related by centro-symmetry.

interior of the gap and the (y, z) flow consists of a single large rotating cell. As is the case for plane Couette flow, variation in z is much slower than variation in y . An obvious difference between the two flows is symmetry: plane Poiseuille flow is reflection-symmetric in y , while plane Couette flow is centro-symmetric in (y, z) . That is, variables u and w obey

$$F(-y, z) = F(y, z) \quad \text{Poiseuille} \quad (11a)$$

$$F(-y, z) = -F(y, -z) \quad \text{Couette} \quad (11b)$$

For other quantities, e.g. v or ψ , the change in sign is opposite to that in (11a) and (11b). One of the striking properties of turbulent-laminar banded patterns is that their mean flow inherits the symmetries of the laminar flow.

Figure 9a shows the main forces acting in the streamwise direction on the mean flow:

$$F_{\text{turb}} \equiv -\langle (\tilde{\mathbf{u}} \cdot \nabla) \tilde{\mathbf{u}} \rangle \quad (12a)$$

$$F_{\text{adv}} \equiv -\langle \mathbf{u}_{\text{lam}} \cdot \nabla \rangle \langle u - u_{\text{lam}} \rangle \quad (12b)$$

$$F_{\text{visc}} \equiv \frac{1}{Re} \nabla^2 \langle u - u_{\text{lam}} \rangle \quad (12c)$$

We omit the larger forces governing laminar Poiseuille flow:

$$\nabla p_{\text{lam}} = \frac{1}{Re} \nabla^2 \mathbf{u}_{\text{lam}} = -\frac{2}{Re} \mathbf{e}_{\text{strm}} \quad (13)$$

as well as the smaller pressure gradient associated with $\langle \mathbf{u} - \mathbf{u}_{\text{lam}} \rangle$ and the nonlinear interaction of $\langle \mathbf{u} - \mathbf{u}_{\text{lam}} \rangle$ with itself. The three forces are plotted as a function of z for values of y near the two walls and at the center of the channel. To interpret Fig. 9, it is helpful to recall that while z is not the streamwise direction, it has a component in this direction; see Fig. 1c. Thus a streamwise force which is positive (negative) accelerates (decelerates) the fluid towards the right (left) in the z direction.

Figure 9b shows the balance of forces for an analogous state in plane Couette flow. We have chosen $Re = 300$ because at this Reynolds number, the turbulence in plane Couette flow is localized⁸, as is the case for plane Poiseuille flow at $Re = 1100$. The domain of length $L_z = 80$ is chosen to be twice that of the domain we used for Poiseuille flow. For each flow, the origin in z has been translated so that the turbulent region is at the center of the graph. The nonzero y positions for which the forces are plotted are those for which the forces are maximal. The different symmetries of the two flows, i.e. y -reflection for Poiseuille and centro-symmetry in (y, z) for Couette flow, are clearly visible in Fig. 9

The upper panels show the forces for $y < 0$, where the shear of both basic flows is positive. The resemblance between the force balance for the two flows is remarkable. For both flows, F_{turb} is large and positive in the turbulent region, accelerating the fluid towards the right, and is counterbalanced primarily by F_{visc} . The advective force F_{adv} is comparable but smaller in magnitude; it acts with F_{turb} in the left portion of the turbulent region and

against F_{turb} in the right portion. At the center ($y = 0$), the curvature $\partial_y^2 u$ and hence F_{visc} is small for Poiseuille flow and negligible for Couette flow. For Poiseuille flow, F_{adv} accelerates the fluid in the turbulent region, while F_{turb} decelerates it. For Couette flow, this holds over half the turbulent region, while the reverse is true over the other half, as required by (11b). For $y > 0$, the flow and forces for Poiseuille flow are the same as for $y < 0$, while for Couette flow the flow and forces are reversed from those at $y < 0$.

IV. DISCUSSION

We conclude with some further comparisons between turbulent-laminar bands in Poiseuille and Couette flow. The similarity in the balance of forces shown in figure 9 also strongly supports the idea that turbulent-laminar patterns are maintained by the same physical mechanisms in Poiseuille and Couette flow. It has been proposed by Waleffe²⁴ that plane Poiseuille flow can be viewed as two superposed plane Couette flows. This is consistent with the two cells and two turbulent regions seen in Fig. 8d,e.

With this in mind, we compare the wavelengths and Reynolds numbers of turbulent-laminar banded patterns in Poiseuille and Couette flow. In our domain, the wavelength of the patterns in plane Poiseuille flow is 20 at higher Re and becomes 40 for lower Re . Turbulent-laminar patterns in plane Couette flow have higher wavelengths⁶⁻¹⁰: 40 for higher Re and 60 for lower Re . The idea of considering Poiseuille flow as two superposed Couette flows suggests that Poiseuille flow should be scaled by the quarter-gap rather than the half-gap. This would make the pattern wavelength of 40 (quarter-gaps) of plane Poiseuille flow at higher Re consistent with the wavelength of 40 (half-gaps) observed for plane and Taylor-Couette flow. It was already observed^{7,9} that a unified Reynolds number Re_s based on the square of the y -averaged shear of the laminar flow and the quarter gap for plane Poiseuille flow could be defined to yield $Re_s = Re_c/4 \approx Re/4.6$. For plane Couette flow, using the constant shear and the half-gap, Re_s is the usual Reynolds number. The range of existence $800 \leq Re \leq 1900$ for turbulent-laminar patterns in plane Poiseuille flow becomes $174 \leq Re_s \leq 413$, which is of the same order as the range of existence $[300, 420]$ for turbulent-laminar patterns in plane Couette flow. We note that the Reynolds-number range over which patterned turbulence is obtained numerically by Tsukahara et al.¹⁸ is $Re_m \in [1125, 2250]$, i.e. $Re_s \in [245, 490]$ and the range obtained experimentally by Hashimoto et al.²⁰ is $Re_m \in [1275, 1500]$, i.e. $Re_s \in [277, 326]$. It would be unlikely to obtain more precise agreement since the analogy between Poiseuille and Couette flow is inexact in a number of ways. For example, the turbulent near-wall regions of plane Poiseuille flow occupy considerably less than a quarter-gap. Second, in seeking a single measure

of the shear in plane Poiseuille flow, it is not clear that a simple average is the best candidate.

An important consequence of the differences in symmetry is that the moderate-time averages of the turbulent-laminar patterns in Couette flow are stationary while those of Poiseuille flow have a well-defined velocity, which we have shown in Fig. 4.

It is almost surely possible to produce patterns at angles quite different from 24° . In large-scale experiments in plane Couette flow^{6,7} patterns were observed whose angles ranged between 24° and 37° ; simulations in narrow tilted domains with imposed angles ranging from 15° and 66° all produced patterns^{8–10}. The simulated patterns whose angles are far outside the range [$24^\circ, 37^\circ$] would presumably be unstable when placed in a less constrained geometry. Although the narrow tilted geometry – the analogue of the minimal flow unit^{22,23} for maintaining shear-flow turbulence – can be used to study some of the characteristics of turbulent-laminar patterns, studies in a less constrained geometry are necessary for understanding their genesis and fate. The spreading of turbulent spots and fronts have been widely studied for plane Poiseuille flow, e.g. experimentally by Lemoult et al.^{33,34} and numerically by Aida et al.³⁵ and as well as numerically by Duguet et al.³⁶ for plane Couette flow

Future work will focus on the mechanism maintaining turbulent-laminar patterns and on the branching events that accompany the change in wavelength and in speed.

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